## **ICON Data Product 2.7: IVM Ionospheric Parameters**

This document describes the data product for IVM L2 Thermal Plasma file, which is in NetCDF4 format.

This the version 5 release of geophysical data for the Ion Velocity Meter (IVM) on-board the Ionospheric Connections Explorer (ICON) satellite. The IVM is comprised of two instruments, the Retarding Potential Analyzer (RPA) and the Drift Meter (DM). The IVM is operating very well and with Iow noise, generally producing high quality outputs. Current work on the data processing is driven by the generally low ionospheric densities resulting from Iow solar activity. During daytime hours, particularly mid-morning and later, all parameters are generally available and high quality.

This product contains several data quality flags that should be examined to ensure the highest fidelity data. Good data for all parameters, including ion drifts, may be down-selected by selecting times where two flags,ICON\_L27\_RPA\_FLAG and ICON\_L27\_DM\_FLAG, are both zero and IVM\_L27\_Attitude\_Status is less than 8. Parameters such as ion density and composition are generally always reliable, followed by ion temperature, then the three dimensional ion drift vector. The ion temperature may demonstrate an increased variance in the early morning hours but early analysis indicates the values are otherwise high quality. Unlike the other parameters, determination of the ion drifts requires a minimum absolute O+ density. At night, the absolute O+ densities can fall below the detection limit of the IVM. When this occurs, resolving the ion motion along the IVM look direction can be challenging. During these periods the ion velocity is set to the co-rotation velocity. Use of the full 3D vectors (ICON\_L27\_Ion\_Velocity\_Meridional, ICON\_L27\_Ion\_Velocity\_Zonal, and ICON L27 Ion Velocity Field Aligned) during these periods is not recommended (ICON L27 RPA FLAG=1). The magnetic meridional ion drift near the magnetic equator, magnetic latitudes +/- 5 degrees, is primarily given by ICON L27 Ion Velocity Z (positive towards Earth), and may be used as an approximation even when the RPA\_FLAG=1. The Drift Meter (DM) measures cross-track motions and excludes H+ ions. Low O+ densities, particularly at dawn, make measurements a challenge. A flag is set when O+ densities are too low for the hardware to function (ICON\_L27\_DM\_FLAG=3). The low O+ densities exacerbates the impact of photoemission upon measurements within the DM. A first-order model has been developed and work is progressing on producing high-quality outputs during these periods. In the interim, ion drifts that require a significant photoemission correction have been masked (ICON\_L27\_DM\_FLAG=2). Any spacecraft operations that may have the potential to impact the outputs are currently flagged (ICON L27 RPA FLAG=2, ICON\_L27\_DM\_FLAG=1). Impacts are likely more prevalent in the DM than RPA, depending upon the operation. See spacecraft flags for more.

# History

Version 001, R. A. Stoneback, 2019-08-06T00:00:00, Initial Release

Version 002, R. A. Stoneback, 2020-06-19T00:00:00, Update for public release, adds quality flags

Version 003, R. A. Stoneback, 2020-04-10T00:00:00, Adds first order photoemission correction

Version 004, R. A. Stoneback, 2020-12-03T00:00:00, Adjustments to IVM offsets

Version 005, R. A. Stoneback, 2021-03-21T00:00:00, Improvements to RPA Fit noise, Drifts in East, North, Up. Data between 0500 and 1200 should be rejected at this time due to poor S/N and contaminant signals from internally generated photocurrents that have not been removed.

The V05 data product has been corrected for long-term systematic offsets produced by uncertainties in the electrostatic environment of the spacecraft. The corrected variables are:

ICON\_L27\_Ion\_Velocity\_X

ICON\_L27\_Ion\_Velocity\_Y

ICON\_L27\_Ion\_Velocity\_Z

They are used to compute the plasma drifts in magnetic coordinates.

The uncorrected data may be accessed directly in the variables

ICON\_L27\_Raw\_Ion\_Velocity\_X

ICON\_L27\_Raw\_Ion\_Velocity\_Y

#### ICON\_L27\_Raw\_Ion\_Velocity\_Z

Short term variations with periods less than 10 days have not been removed but may be accurately assessed from examination of the zonal (daily) average of the meridional drift ICON\_L27\_lon\_Velocity\_Meridional within 1 hour of 1800 MLT. Zonal averages in excess of 5 m/s over this local time range provide a reliable estimate of the short-term offset.

## **Dimensions**

NetCDF files contain **variables** and the **dimensions** over which those variables are defined. First, the dimensions are defined, then all variables in the file are described.

The dimensions used by the variables in this file are given below, along with nominal sizes. Note that the size may vary from file to file. For example, the "Epoch" dimension, which describes the number of time samples contained in this file, will have a varying size.

Dimension Name	Nominal Size
Epoch	unlimited

# Variables

Variables in this file are listed below. First, "data" variables are described, followed by the "support\_data" variables, and finally the "metadata" variables. The variables classified as "ignore\_data" are not shown.

#### data

Variable Name	Description	Units	Dimensions
Epoch	Universal Time (UTC) Time at the midpoint of the IVM measurements.	Millisec onds since 1 970-01- 01 00:0 0:00	Epoch
ICON_L27_Altitude	WGS84 Altitude of Spacecraft Position (geodetic)	km	Epoch
	Geodetic Altitude of Spacecraft in WGS84.		
ICON_L27_Fractional	Fraction of total plasma number density that is H+		Epoch
_ION_DENSITY_H	Determined via a non-linear least squares fit of RPA currents to the Whipple equation		
ICON_L27_Fractional	Fraction of total plasma number density that is O+		Epoch
_ION_DENSITY_0	Determined via a non-linear least squares fit of RPA currents to the Whipple equation		
ICON_L27_Latitude	WGS84 Latitude of Spacecraft Position (geodetic)	degree	Epoch
	Geodetic latitude of spacecraft in WGS84	SINOIT	
ICON_L27_Solar_Loca	Local Solar Time at Spacecraft	hour	Epoch
	Local Solar Time at spacecraft.		
ICON_L27_Magnetic_L	Magnetic Local Time at Spacecraft	hour	Epoch
ocal_lime	Magnetic Local Time at the spacecraft.		
ICON_L27_Longitude	WGS84 Longitude of Spacecraft Position (geodetic)	degree	Epoch
	Geodetic longitude of spacecraft in WGS84	5 2051	
ICON_L27_Ion_Densit	Ion density determined using RPA measurements.	N/cc	Epoch
У	lon density uses measured currents and co-rotating atmosphere to determine density.		
ICON_L27_Ion_Temper ature	Ion temperature determined using a best fit of RPA measurements to Whipple equation.	К	Epoch
	Temperature is obtained by assuming single temperature value for all plasma.		

Variable Name	Description	Units	Dimensions
ICON_L27_Ion_Veloci ty_Field_Aligned	Field-Aligned Ion Velocity Ion velocity relative to co-rotation along geomagnetic field lines. Positive along the main field vector.	m/s	Epoch
	Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame to express the observed vector along a geomagnetic basis.		
ICON_L27_Ion_Veloci	Meridional Ion Velocity	m/s	Epoch
ty_Meridional	Ion velocity along local magnetic meridional direction, perpendicular to geomagnetic field and within local magnetic meridional plane. The local meridional vector maps to vertical at the magnetic equator, positive is up.		
	Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame to express the observed vector along a geomagnetic basis.		
ICON_L27_Ion_Veloci	Ion velocity along IVM-x	m/s	Epoch
	Cross-track velocity is relative to co-rotation and in the instrument frame. Positive-x is normal to IVM aperture plane and in the direction of satellite motion. Velocity obtained through fitting of RPA currents to the Whipple equation to get a measure of the total along track ion velocity as observed within the instrument. Signals produced by the motion of the spacecraft and the rotation of the Earth are removed to produce this result. Mean offset of 66.5 added by R. A. Heelis.		
ICON_L27_Ion_Veloci	Ion velocity along IVM-y	m/s	Epoch
	Cross-track velocity is relative to co-rotation and in the instrument frame. Positive-y points generally southward when the instrument is pointed along the ram direction. Velocity obtained through conversion of arrival angles measured by the DM into a cross track velocity using trigonometry. Signals produced by the motion of the spacecraft and the rotation of the Earth are removed to produce this result. Mean offset of -0.0 added by R. A. Heelis.		
ICON_L27_Ion_Veloci	Ion velocity along IVM-z	m/s	Epoch
	Cross-track velocity is relative to co-rotation and in the instrument frame. Positive-z is directed towards nadir (Earth). Velocity obtained through conversion of arrival angles measured by the DM into a cross track velocity using trigonometry. Signals produced by the motion of the spacecraft and the rotation of the Earth are removed to produce this result. Mean offset of 1.5 added by R. A. Heelis.		

Variable Name	Description	Units	Dimensions
ICON_L27_Ion_Veloci ty_Zonal	Zonal ion velocity	m/s	Epoch
	Ion velocity relative to co-rotation along the magnetic zonal direction, normal to local magnetic meridional plane and the geomagnetic field (positive east). The local zonal vector maps to purely horizontal at the magnetic equator. Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame to express the observed vector along a geomagnetic basis.		
ICON_L27_Raw_Ion_Ve locity_X	Raw ion velocity determined using a best fit of RPA measurements to the Whipple equation.	m/s	Epoch
	This is the total ion velocity along normal direction into the RPA, including s/c motion.		
ICON_L27_Raw_Ion_Ve locity_Y	Total ion velocity measured along IVM-y.	m/s	Epoch
	Total observed ion velocity along instrument cross-track direction, reported in the instrument frame. The translation of DM measured angles to ion velocities uses knowledge of the ram velocity of the plasma and the electric potential of the instrument aperture relative to the ambient plasma, both of which are provided by the RPA.		
ICON_L27_Raw_Ion_Ve locity_Z	Total ion velocity measured along IVM-z.	m/s	Epoch
	Total observed ion velocity along instrument cross-track direction, reported in the instrument frame. The translation of DM measured angles to ion velocities uses knowledge of the ram velocity of the plasma and the electric potential of the instrument aperture relative to the ambient plasma, both of which are provided by the RPA.		
ICON_L27_Unit_Vecto r Meridional X	Unit vector for the geomagnetic meridional direction.		Epoch
	The meridional unit vector is perpendicular to the geomagnetic field and maps along magnetic field lines to vertical at the magnetic equator, where positive is up. The unit vector is expressed here in the IVM coordinate system, where x is along the IVM boresight, nominally along ram when in standard pointing. Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		
ICON_L27_Unit_Vecto r_Meridional_Y	Unit vector for the geomagnetic meridional direction.		Epoch
	The meridional unit vector is perpendicular to the geomagnetic field and maps along magnetic field lines to vertical at the magnetic equator, where positive is up. The unit vector is expressed here in the IVM coordinate system, where $Y = Z \times X$ , nominally southward when in standard pointing, X along ram. Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		

Variable Name	Description	Units	Dimensions
ICON_L27_Unit_Vecto r_Meridional_Z	Unit vector for the geomagnetic meridional direction. The meridional unit vector is perpendicular to the geomagnetic field and maps along magnetic field lines to vertical at the magnetic equator, where positive is up. The unit vector is expressed here in the IVM coordinate system, where Z is nadir pointing (towards Earth), when in standard pointing, X along ram. Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		Epoch
ICON_L27_IVM_Apertu re_Potential	Plasma potential relative to instrument aperture plane - determined using a best fit to RPA measurements. Incident plasma will have some potential to the IVM aperture plane. The aperture plane voltage matches that of a conductor allowed to float electrically with respect to the spacecraft. The flux of ions (driven by s/c motion) must be balanced by the flux of electrons (driven by electron temperature). The value of the aperture plane potential evolves naturally to limit the collection of electrons such the net flux is zero.	V	Epoch
ICON_L27_Unit_Vecto r_Field_Aligned_X	Unit vector for the geomagnetic field line direction. The field-aligned vector points along the geomagnetic field, with positive values along the field direction, and is expressed here in the IVM instrument frame. The IVM-x direction points along the instrument boresight, which is pointed into ram for standard operations.Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		Epoch
ICON_L27_Unit_Vecto r_Field_Aligned_Y	Unit vector for the geomagnetic field line direction. The field-aligned vector points along the geomagnetic field, with positive values along the field direction. The unit vector is expressed here in the IVM coordinate system, where $Y = Z \times X$ , nominally southward when in standard pointing, X along ram. Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		Epoch
ICON_L27_Unit_Vecto r_Field_Aligned_Z	Unit vector for the geomagnetic field line direction. The field-aligned vector points along the geomagnetic field, with positive values along the field direction, and is expressed here in the IVM instrument frame. The IVM-Z direction points towards nadir when IVM-X is pointed into ram for standard operations.Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		Epoch

Variable Name	Description	Units	Dimensions
ICON_L27_Unit_Vecto r_Zonal_X	Unit vector for the zonal geomagnetic direction. The zonal vector is perpendicular to the local magnetic field and the magnetic meridional plane. The zonal vector maps to purely horizontal at the magnetic equator, with positive values pointed generally eastward. This vector is expressed here in the IVM instrument frame.The IVM-x direction points along the instrument boresight, which is pointed into ram for standard operations.Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		Epoch
ICON_L27_Unit_Vecto r_Zonal_Y	Unit vector for the zonal geomagnetic direction. The zonal vector is perpendicular to the local magnetic field and the magnetic meridional plane. The zonal vector maps to purely horizontal at the magnetic equator, with positive values pointed generally eastward. The unit vector is expressed here in the IVM coordinate system, where $Y = Z \times X$ , nominally southward when in standard pointing, X along ram. Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		Epoch
ICON_L27_Unit_Vecto r_Zonal_Z	Unit vector for the zonal geomagnetic direction. The zonal vector is perpendicular to the local magnetic field and the magnetic meridional plane. The zonal vector maps to purely horizontal at the magnetic equator, with positive values pointed generally eastward. This vector is expressed here in the IVM instrument frame.The IVM-Z direction points towards nadir when IVM-X is pointed into ram for standard operations.Calculated using the corresponding unit vector in ECEF and the orientation of the IVM also expressed in ECEF (sc_*hat_*).		Epoch
ICON_L27_MTB_Status	Magnetic Torquer Bar Firing Status If the magnetic torquers are active during any part of the measurement, it is recorded as active for whole measurement. Decoded from spacecraft housekeeping file: ICON_L0_Spacecr aft_Housekeeping-MTB_2020-01-01_v01r001.CSV	binary	Epoch
ICON_L27_Slew_Statu			Epoch
ICON_L27_Sun_Status	Spacecraft Sun/Shadow Status Code Data is from predictive ephemeris. 0 = spacecraft in Sun, 1 = spacecraft in Earth Shadow.	binary	Epoch

Variable Name	Description	Units	Dimensions
ICON_L27_Space_Envi	Space Environment Region Status	binary	Epoch
ronment_Region_stat	Standarized for several missions, not all codes are relevant for		
us	ICON where		
	1: Earth Shadow		
	2: Lupar Shadow		
	A: Atmospheric Absorption Zone		
	8: South Atlantic Anomaly		
	16: Northern Auroral Zone		
	22: Southern Auroral Zone		
	64: Porionsis Passago		
	129: Inner 8 Outer Rediction Polto		
	126. Inner & Outer Radiation Bells		
	230. Deep Plasma Sphere		
	1024. Solar Wind Boom		
	1024: Solar Wind Beam		
	2048: High Magnetic Field		
	4096: Average Plasma Sheet		
	8192: Bowsnock Crossing		
	16384: Magnetopause Crossing		
	32768: Ground Based Observatories		
	65536: 2-Day Conjunctions		
	1310/2: 4-Day Conjunctions		
	262144: Time Based Conjunctions		
	524288: Radial Distance Region 1		
	1048576: Orbit Outbound		
	2097152: Orbit Inbound		
	4194304: Lunar Wake		
	8388608: Magnetotail		
	16777216: Magnetosheath		
	33554432: Science		
	67108864: Low Magnetic Latitude		
	134217728: Conjugate Observation		
ICON_L27_Attitude_S	Slew or Off-Point Status Code	binary	Epoch
tatus	D'anna Orderlate anna h-an		
	Binary Coded Integer where		
	1: LVLH Normal Mode		
	2: LVLH Reverse Mode		
	4: Earth Limb Pointing		
	8: Inertial Pointing		
	16: Stellar Pointing		
	32: Attitude Slew		
	64: Conjugate Maneuver		
	128: Nadir Calibration		
	256: Lunar Calibration		
	512: Stellar Calibration		
	1024: Zero Wind Calibration		
	2048-32768: SPARE		
ICON_L27_Orbit_Numb	Orbit Number	integer	Epoch
	Integer Orbit Number.		

Variable Name	Description	Units	Dimensions
ICON_L27_Magnetic_L atitude	Magnetic Latitude of Spacecraft Position Quasi-dipole magnetic latitude of the spacecraft position. These values are obtained from passing the geodectic latitudes, longitudes, and altitudes from ICON_ANCILLARY_IVM_LATITUDE, ICON_ANCILLARY_IVM_LONGITUDE, and ICON_ANCILLARY_IVM_ALTITUDE into apexpy Python module. For details on apexpy see: https://apexpy.readthedocs.org/	degree s North	Epoch
ICON_L27_Magnetic_L ongitude	Magnetic Longitude of Spacecraft Position Quasi-dipole magnetic longitude of the spacecraft position. These values are obtained from passing the geodectic latitudes, longitudes, and altitudes from ICON_ANCILLARY_IVM_LATITUDE, ICON_ANCILLARY_IVM_LONGITUDE, and ICON_ANCILLARY_IVM_ALTITUDE into apexpy Python module. For details on apexpy see: https://apexpy.readthedocs.org/	degree s East	Epoch
ICON_L27_DM_Flag	Drift meter quality flag. This flag is less than 1 when the data is of highest quality. Vx and magnetic component drifts should be rejected when this flag is greater than 0. All data should rejected when this flag is greater than 1 0 - Good data. 1 - Data may have artifacts due to s/c operations. 2 - Data temporarily removed for photoemission. 3 - Not enough O+ to measure arrival angle.4 - Spacecraft attitude may significantly impact performance.		Epoch
ICON_L27_RPA_Flag	RPA Quality Flag Status flag for RPA. This flag is less than 1 when the data is of highest quality and greater than 1 when it should be rejected. 0 - All RPA parameters are good. Ion temperatures correspond to both O+ and H+. 1 - Ram Ion Velocities are not good. Other parameters good. Plasma is presumed to be co-rotating when fitting RPA curves that have an insufficient quantity of O+. Ion temperatures correspond to H+ only. 2 - Geophysical outputs may be impacted by spacecraft operations. 3 - Geophysical outputs may be significantly impacted by spacecraft attitude.		Epoch

Variable Name	Description	Units	Dimensions
ICON_L27_UTC_Time	ISO 9601 formatted UTC timestamp (at middle of reading).		Epoch
	ISO 9601 formatted UTC timestamp (at middle of reading).		
	Time is generated from the time-code at byte 1015 of the IVM packet minus the time sync at byte 1019 of the IVM packet. This is the GPS time at the start of the integration period. The integration period is assumed to be 4 seconds so the center time is 2 seconds after that. The formula is (time-code * 1000ms) + 2000ms - (16 * time sync / 1000) in GPS milliseconds then converted to UTC time. See the UTD 206-024 Rev A document.		
	Time may be delayed by up to 10 ms due to FSW polling delay.		
	Maximum time is ~2150 UTC and minimum time is ~1970 UTC.		
ICON_L27_Time_UTC_S tart	Milliseconds since 1970-01-01 00:00:00 UTC at start of reading.	millisec onds	Epoch
	Milliseconds since 1970-01-01 00:00:00 UTC at start of reading.		
	Time is generated from the time-code at byte 1015 of the IVM packet minus the time sync at byte 1019 of the IVM packet. This is the GPS time at the start of the integration period. The integration period is assumed to be 4 seconds so the center time is 2 seconds after that. The formula is (time-code * 1000ms) + 2000ms - (16 * time sync / 1000) in GPS milliseconds then converted to UTC time. See the UTD 206-024 Rev A document.		
	Time may be delayed by up to 10 ms due to FSW polling delay.		
	Maximum time is ~2150 UTC and minimum time is ~1970 UTC.		
ICON_L27_Time_UTC_S top	Milliseconds since 1970-01-01 00:00:00 UTC at end of reading.	millisec onds	Epoch
	Milliseconds since 1970-01-01 00:00:00 UTC at end of reading.		
	Time is generated from the time-code at byte 1015 of the IVM packet minus the time sync at byte 1019 of the IVM packet. This is the GPS time at the start of the integration period. The integration period is assumed to be 4 seconds so the center time is 2 seconds after that. The formula is (time-code * 1000ms) + 2000ms - (16 * time sync / 1000) in GPS milliseconds then converted to UTC time. See the UTD 206-024 Rev A document.		
	Time may be delayed by up to 10 ms due to FSW polling delay.		
	Maximum time is ~2150 UTC and minimum time is ~1970 UTC.		
ICON_L27_Apex_Heigh	Modified APEX Height	km	Epoch
	Modified APEX height of the spacecraft position.		

Variable Name	Description	Units	Dimensions
ICON_L27_GPS_Epoch	Milliseconds since 1980-01-06 00:00:00 TAI (coincident with UTC) at middle of reading.	millisec onds	Epoch
	Milliseconds since 1980-01-06 00:00:00 TAI (coincident with UTC) at middle of reading.		
	Time is generated from the time-code at byte 1015 of the IVM packet minus the time sync at byte 1019 of the IVM packet. This is the GPS time at the start of the integration period. The integration period is assumed to be 4 seconds so the center time is 2 seconds after that. The formula is (time-code * 1000ms) + 2000ms - (16 * time sync / 1000) in GPS milliseconds then converted to UTC time. See the UTD 206-024 Rev A document.		
	Time may be delayed by up to 10 ms due to FSW polling delay.		
	Maximum time is ~2150 UTC and minimum time is ~1970 UTC.		
ICON_L27_A_Status	IVM-A Status	binary	Epoch
	<ul> <li>Binary Coded Integer where</li> <li>1: Earth Day View</li> <li>2: Earth Night View</li> <li>4: Calibration Target View</li> <li>8: Off-target View</li> <li>16: Sun Proximity View</li> <li>32: Moon Proximity View</li> <li>32: Moon Proximity View</li> <li>64: North Magnetic Footpoint View</li> <li>128: South Magnetic Footpoint View</li> <li>256: Science Data Collection View</li> <li>512: Calibration Data Collection View</li> <li>1024: RAM Proximity View</li> <li>2048-32768: SPARE</li> </ul>		
	Activity is what the spacecraft was commanded to do while status is the spacecraft's natural state of operations. This means that activity should always be used over status if they differ, but will almost always be the same.		

Variable Name	Description	Units	Dimensions
ICON_L27_B_Status	IVM-B Status	binary	Epoch
	Binary Coded Integer where 1: Earth Day View 2: Earth Night View 4: Calibration Target View 8: Off-target View 16: Sun Proximity View 16: Sun Proximity View 32: Moon Proximity View 64: North Magnetic Footpoint View 128: South Magnetic Footpoint View 128: South Magnetic Footpoint View 128: South Magnetic Footpoint View 256: Science Data Collection View 512: Calibration Data Collection View 1024: RAM Proximity View 2048-32768: SPARE Activity is what the spacecraft was commanded to do while status is the spacecraft's natural state of operations. This means that activity should always be used over status if they differ but will element olympus to the same		
TCON L27 A Activity	IVM-A Activity	binary	Epoch
	<ul> <li>Binary Coded Integer where:</li> <li>1: Earth Day View</li> <li>2: Earth Night View</li> <li>4: Calibration Target View</li> <li>8: Off-target View</li> <li>16: Sun Proximity View</li> <li>32: Moon Proximity View</li> <li>64: North Magnetic Footpoint View</li> <li>128: South Magnetic Footpoint View</li> <li>256: Science Data Collection View</li> <li>512: Calibration Data Collection View</li> <li>1024: RAM Proximity View</li> <li>2048-32768: SPARE</li> </ul>		
	Activity is what the spacecraft was commanded to do while status is the spacecraft's natural state of operations. This means that activity should always be used over status if they differ, but will almost always be the same.		

Variable Name	Description	Units	Dimensions
ICON_L27_B_Activity	IVM-A Activity	binary	Epoch
	Binary Coded Integer where:		
	1: Forth Day View		
	2: Earth Night View		
	2. Earth Night View		
	4: Calibration Target View		
	8: Off-target View		
	16: Sun Proximity View		
	32: Moon Proximity View		
	128: South Magnetic Footpoint View		
	256: Science Data Collection View		
	512: Calibration Data Collection View		
	1024: RAM Proximity View		
	2048-32768: SPARE		
	Activity is what the spacecraft was commanded to do while		
	status is the spacecraft's natural state of operations. This		
	means that activity should always be used over status if they		
	differ, but will almost always be the same.		
ICON L27 Footpoint	Altitude of North Footpoint of Geomagnetic Line at 150	km	Epoch
Altitude_North	km from IGRF		
	Altitude location of the magnetic footpoint in the Northern		
	Hemisphere at 150 km. These data were interpolated using a		
	tricubic algorithm from IGRF and ephemeris data then linearly		
	interploted to IVM times.		
ICON_L27_Footpoint_	ECEF X-Component of Field Aligned Drift Direction at	dimensi	Epoch
Field_Aligned_Vecto	Northern Footpoint	onless	
r_ECEF_X_North			
	At the northern footpoint this is the x-component of the unit		
	vector for field aligned ion drifts expressed in the ECEF frame.		
ICON L27 Footpoint	ECEF Y-Component of Field Aligned Drift Direction at	dimensi	Epoch
Field_Aligned_Vecto	Northern Footpoint	onless	-
r_ECEF_Y_North			
	At the northern footpoint this is the y-component of the unit		
	vector for field aligned ion drifts expressed in the ECEF frame.		
ICON L27 Footpoint	ECEF Z-Component of Field Aligned Drift Direction at	dimensi	Epoch
Field_Aligned_Vecto	Northern Footpoint	onless	-
r_ECEF_Z_North			
	At the northern footpoint this is the z-component of the unit		
	vector for field aligned ion drifts expressed in the ECEF frame.		
ICON_L27_Footpoint	Latitude of North Footpoint of Geomagnetic Line at 150	degree	Epoch
Latitude North	km from IGRF	s	-
	Latitude location of the magnetic footpoint in the Northern		
	Hemisphere at 150 km. These data were interpolated using a		
	tricubic algorithm from IGRF and ephemeris data then linearly		
	interploted to IVM times.		

Variable Name	Description	Units	Dimensions
ICON_L27_Footpoint_ Longitude_North	Longitude of North Footpoint of Geomagnetic Line at 150 km from IGRF	degree s	Epoch
	Longitude location of the magnetic footpoint in the Northern Hemisphere at 150 km. These data were interpolated using a tricubic algorithm from IGRF and ephemeris data then linearly interploted to IVM times.		
ICON_L27_Footpoint_ Meridional_Ion_Velo city_North	Northern Meridional Ion Velocity Velocity along local magnetic meridional direction, perpendicular to geomagnetic field and within local magnetic meridional plane, field-line mapped to northern footpoint. The meridional vector is purely vertical at the magnetic equator, positive up. Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame. Field-line mapping and the assumption of equi-potential field lines is used to translate the locally measured ion motion to the magnetic footpoint. The mapping is used to determine the change in magnetic field line distance, which, under assumption of equipotential field lines, in turn alters the electric field at that	m/s	Epoch
	location (E=V/d).		
ICON_L27_Footpoint_ Magnetic_Latitude_N orth	Quasi-dipole Latitude of Northern Footpoint Calculated value of quasi-dipole latitude of northern footpoint from IGRF.	degree s	Epoch
ICON_L27_Footpoint_ Magnetic_Longitude_ North	Quasi-dipole Longitude of Northern Footpoint Calculated value of quasi-dipole longitude of northern footpoint from IGRF	degree s	Epoch
ICON_L27_Footpoint_ Meridional_Vector_E CEF_X_North	ECEF X-Component of Meridional Drift Direction at Northern Footpoint At the northern footpoint this is the x-component of the unit	dimensi onless	Epoch
ICON_L27_Footpoint_ Meridional_Vector_E CEF_Y_North	vector for meridional ion drifts expressed in the ECEF frame. ECEF Y-Component of Meridional Drift Direction at Northern Footpoint	dimensi onless	Epoch
	vector for meridional ion drifts expressed in the ECEF frame.		
ICON_L27_Footpoint_ Meridional_Vector_E CEF_Z_North	ECEF Z-Component of Meridional Drift Direction at Northern Footpoint At the northern footpoint this is the z-component of the unit vector for meridional ion drifts expressed in the ECEF frame.	dimensi onless	Epoch

Variable Name	Description	Units	Dimensions
ICON_L27_Footpoint_ Zonal_Ion_Velocity_	Northern Zonal Ion Velocity	m/s	Epoch
North	Velocity along local magnetic zonal direction, perpendicular to geomagnetic field and the local magnetic meridional plane, field-line mapped to northern footpoint. The zonal vector is purely horizontal when mapped to the magnetic equator, positive is generally eastward. Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame. Field-line mapping and the assumption of equi-potential field lines is used to translate the locally measured ion motion to the northern footpoint. The mapping is used to determine the change in magnetic field line distance, which, under assumption of equipotential field lines, in turn alters the electric field at that		
	location (E=V/d).		
ICON_L27_Footpoint_ Zonal_Vector_ECEF_X	ECEF X-Component of Zonal Drift Directrion at Northern Footpoint	dimensi onless	Epoch
_North	At the northern footpoint this is the x-component of the unit vector for zonal ion drifts expressed in the ECEF frame.		
ICON_L27_Footpoint_ Zonal_Vector_ECEF_Y	ECEF Y-Component of Zonal Drift Directrion at Northern Footpoint	dimensi onless	Epoch
	At the northern footpoint this is the y-component of the unit vector for zonal ion drifts expressed in the ECEF frame.		
ICON_L27_Footpoint_ Zonal_Vector_ECEF_Z	ECEF Z-Component of Zonal Drift Direction at Northern Footpoint	dimensi onless	Epoch
	At the northern footpoint this is the z-component of the unit vector for zonal ion drifts expressed in the ECEF frame.		
ICON_L27_Footpoint_ Altitude_South	Altitude of South Footpoint of Geomagnetic Line at 150 km from IGRF	km	Epoch
	Altitude location of the magnetic footpoint in the Northern Hemisphere at 150 km. These data were interpolated using a tricubic algorithm from IGRF and ephemeris data then linearly interploted to IVM times.		
ICON_L27_Footpoint_ Field_Aligned_Vecto r_ECEF_X_South	ECEF X-Component of Field Aligned Drift Direction at Southern Footpoint	dimensi onless	Epoch
	At the Southern footpoint this is the x-component of the unit vector for field aligned ion drifts expressed in the ECEF frame.		
ICON_L27_Footpoint_ Field_Aligned_Vecto r_ECEF_Y_South	ECEF Y-Component of Field Aligned Drift Direction at Southern Footpoint	dimensi onless	Epoch
	At the Southern footpoint this is the y-component of the unit vector for field aligned ion drifts expressed in the ECEF frame.		
ICON_L27_Footpoint_ Field_Aligned_Vecto r_ECEF_Z_South	ECEF Z-Component of Field Aligned Drift Direction at Southern Footpoint	dimensi onless	Epoch
	At the Southern footpoint this is the z-component of the unit vector for field aligned ion drifts expressed in the ECEF frame.		

Variable Name	Description	Units	Dimensions
ICON_L27_Footpoint_ Latitude_South	Latitude of South Footpoint of Geomagnetic Line at 150 km from IGRF	degree s	Epoch
	Latitude location of the magnetic footpoint in the Southern Hemisphere at 150 km. These data were interpolated using a tricubic algorithm from IGRF and ephemeris data then linearly interploted to IVM times.		
ICON_L27_Footpoint_ Longitude_South	Longitude of South Footpoint of Geomagnetic Line at 150 km from IGRF	degree s	Epoch
	Longitude location of the magnetic footpoint in the Southern Hemisphere at 150 km. These data were interpolated using a tricubic algorithm from IGRF and ephemeris data then linearly interploted to IVM times.		
ICON_L27_Footpoint_ Meridional_Ion_Velo city_South	Southern Meridional Ion Velocity Velocity along local magnetic meridional direction, perpendicular to geomagnetic field and within local magnetic meridional plane, field-line mapped to southern footpoint. The meridional vector is purely vertical at the magnetic equator, positive up. Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame. Field-line mapping and the assumption of equi-potential field lines is used to translate the locally measured ion motion to the magnetic footpoint. The mapping is used to determine the change in magnetic field line distance, which, under assumption of equipotential field lines, in turn alters the electric field at that location (E=V/d).	m/s	Epoch
ICON_L27_Footpoint_ Magnetic_Latitude_S outh	Quasi-dipole Latitude of Southern Footpoint Calculated value of quasi-dipole latitude of southern footpoint from IGRF	degree s	Epoch
ICON_L27_Footpoint_ Magnetic_Longitude_ South	Quasi-dipole Longitude of Southern Footpoint Calculated value of quasi-dipole longitude of southern footpoint from IGRF	degree s	Epoch
ICON_L27_Footpoint_ Meridional_Vector_E CEF_X_South	ECEF X-Component of Meridional Drift Direction at Southern Footpoint At the Southern footpoint this is the x-component of the unit vector for meridional ion drifts expressed in the ECEF frame.	dimensi onless	Epoch
ICON_L27_Footpoint_ Meridional_Vector_E CEF_Y_South	ECEF Y-Component of Meridional Drift Direction at Southern Footpoint At the Southern footpoint this is the y-component of the unit vector for meridional ion drifts expressed in the ECEF frame.	dimensi onless	Epoch
ICON_L27_Footpoint_ Meridional_Vector_E CEF_Z_South	ECEF Z-Component of Meridional Drift Direction at Southern Footpoint At the Southern footpoint this is the z-component of the unit	dimensi onless	Epoch
	vector for meridional ion drifts expressed in the ECEF frame.		

Variable Name	Description	Units	Dimensions
ICON_L27_Footpoint_ Zonal_Ion_Velocity_ South	Southern Zonal Ion Velocity Velocity along local magnetic zonal direction, perpendicular to geomagnetic field and the local magnetic meridional plane, field-line mapped to southern footpoint. The zonal vector is purely horizontal when mapped to the magnetic equator, positive is generally eastward. Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame. Field-line mapping and the assumption of equi-potential field lines is used to translate the locally measured ion motion to the southern footpoint. The mapping is used to determine the change in magnetic field line distance, which, under assumption of equipotential field lines, in turn alters the electric field at that location (E=V/d).	m/s	Epoch
ICON_L27_Footpoint_ Zonal_Vector_ECEF_X _South	ECEF X-Component of Zonal Drift Direction at Southern Footpoint At the Southern footpoint this is the x-component of the unit vector for zonal ion drifts expressed in the ECEF frame.	dimensi onless	Epoch
ICON_L27_Footpoint_ Zonal_Vector_ECEF_Y _South	ECEF Y-Component of Zonal Drift Direction at Southern Footpoint At the Southern footpoint this is the y-component of the unit vector for zonal ion drifts expressed in the ECEF frame.	dimensi onless	Epoch
ICON_L27_Footpoint_ Zonal_Vector_ECEF_Z _South	ECEF Z-Component of Zonal Drift Direction at Southern Footpoint At the Southern footpoint this is the z-component of the unit vector for zonal ion drifts expressed in the ECEF frame.	dimensi onless	Epoch
ICON_L27_Observator y_Magnetic_Field_X	X Component of the Magnetic Field at the Spacecraft X-component of the magnetic field from IGRF at the spacecarft position, expressed in the ECEF frame.	nT	Epoch
ICON_L27_Observator y_Magnetic_Field_Y	Y Component of the Magnetic Field at the Spacecraft Y-component of the magnetic field from IGRF at the spacecarft position, expressed in the ECEF frame.	nT	Epoch
ICON_L27_Observator y_Magnetic_Field_Z	Z Component of the Magnetic Field at the Spacecraft Z-component of the magnetic field from IGRF at the spacecarft position, expressed in the ECEF frame.	nT	Epoch

Variable Name	Description	Units	Dimensions
ICON_L27_Equator_Io n_Velocity_Meridion al	Equatorial Meridional Ion Velocity Velocity along local magnetic meridional direction, perpendicular to geomagnetic field and within local magnetic meridional plane, field-line mapped to apex/magnetic equator. The meridional vector is purely vertical at the magnetic equator, positive up. Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame. Field-line mapping and the assumption of equi-potential field lines is used to translate the locally measured ion motion to the magnetic equator. The mapping is used to determine the change in magnetic field line distance, which, under assumption of equipotential field lines, in turn alters the electric field at that location (E=V/d).	m/s	Epoch
ICON_L27_Equator_Io n_Velocity_Zonal	Equatorial Zonal Ion Velocity Velocity along local magnetic zonal direction, perpendicular to geomagnetic field and the local magnetic meridional plane, field-line mapped to apex/magnetic equator. The zonal vector is purely horizontal when mapped to the magnetic equator, positive is generally eastward. Velocity obtained using ion velocities relative to co-rotation in the instrument frame along with the corresponding unit vectors expressed in the instrument frame. Field-line mapping and the assumption of equi-potential field lines is used to translate the locally measured ion motion to the magnetic equator. The mapping is used to determine the change in magnetic field line distance, which, under assumption of equipotential field lines, in turn alters the electric field at that location (E=V/d).	m/s	Epoch
ICON_L27_Observator y_Velocity_X	ECI Spacecraft Velocity Velocity of spacecraft in ECI, J2000, cooridinates.	m/s	Epoch
ICON_L27_Observator y_Velocity_Y	ECI Spacecraft Velocity Velocity of spacecraft in ECI, J2000, cooridinates.	m/s	Epoch
ICON_L27_Observator y_Velocity_Z	ECI Spacecraft Velocity Velocity of spacecraft in ECI, J2000, cooridinates.	m/s	Epoch
ICON_L27_Observator y_Corotation_X	ECI Earth Corotation Velocity Components in IVM Coordinates Component of Earth's corotation velocity vector projected into the IVM instrument axes by taking the dot product of the corotation vector with the instrument's axes and multiplying the Y and Z components by negative 1 (but not the X component by convention).	m/s	Epoch

Variable Name	Description	Units	Dimensions
ICON_L27_Observator y_Corotation_Y	ECI Earth Corotation Velocity Components in IVM Coordinates	m/s	Epoch
	Component of Earth's corotation velocity vector projected into the IVM instrument axes by taking the dot product of the corotation vector with the instrument's axes and multiplying the Y and Z components by negative 1 (but not the X component by convention).		
ICON_L27_Observator y_Corotation_Z	ECI Earth Corotation Velocity Components in IVM Coordinates	m/s	Epoch
	Component of Earth's corotation velocity vector projected into the IVM instrument axes by taking the dot product of the corotation vector with the instrument's axes and multiplying the Y and Z components by negative 1 (but not the X component by convention).		
ICON_L27_Solar_Zeni th_Angle	Solar Zenith Angle at Spacecraft	degree s	Epoch
	Solar Zenith Angle at the spacecraft.		
ICON_L27_Ion_Veloci ty_East	IVM measured ion drifts along East, North, and Up directions.	m/s	Epoch
ICON_L27_Ion_Veloci ty_North	IVM measured ion drifts along East, North, and Up directions.	m/s	Epoch
ICON_L27_Ion_Veloci ty_Up	IVM measured ion drifts along East, North, and Up directions.	m/s	Epoch
ICON_L27_Unit_Vecto r_X_East	Basis vector system along East, North, and Up directions for IVM-x axis.		Epoch
ICON_L27_Unit_Vecto r_X_North	Basis vector system along East, North, and Up directions for IVM-x axis.		Epoch
ICON_L27_Unit_Vecto r_X_Up	Basis vector system along East, North, and Up directions for IVM-x axis.		Epoch
ICON_L27_Unit_Vecto r_Y_East	Basis vector system along East, North, and Up directions for IVM-y axis.		Epoch
ICON_L27_Unit_Vecto r_Y_North	Basis vector system along East, North, and Up directions for IVM-y axis.		Epoch
ICON_L27_Unit_Vecto r_Y_Up	Basis vector system along East, North, and Up directions for IVM-y axis.		Epoch
ICON_L27_Unit_Vecto r_Z_East	Basis vector system along East, North, and Up directions for IVM-z axis.		Epoch

Variable Name	Description	Units	Dimensions
ICON_L27_Unit_Vecto r_Z_North	Basis vector system along East, North, and Up directions for IVM-z axis.		Epoch
ICON_L27_Unit_Vecto r_Z_Up	Basis vector system along East, North, and Up directions for IVM-z axis.		Epoch
ICON_L27_Footpoint_ East_Ion_Velocity_S outh	Ion drifts at southern footpoint along East, North, and Up directions.	m/s	Epoch
ICON_L27_Footpoint_ North_Ion_Velocity_ South	Ion drifts at southern footpoint along East, North, and Up directions.	m/s	Epoch
ICON_L27_Footpoint_ Up_Ion_Velocity_Sou th	Ion drifts at southern footpoint along East, North, and Up directions.	m/s	Epoch
ICON_L27_Footpoint_ East_Ion_Velocity_N orth	Ion drifts at northern footpoint along East, North, and Up directions.	m/s	Epoch
ICON_L27_Footpoint_ North_Ion_Velocity_ North	Ion drifts at northern footpoint along East, North, and Up directions.	m/s	Epoch
ICON_L27_Footpoint_ Up_Ion_Velocity_Nor th	Ion drifts at northern footpoint along East, North, and Up directions.	m/s	Epoch

### Acknowledgement

This is a data product from the NASA lonospheric Connection Explorer mission, an Explorer launched at 21:59:45 EDT on October 10, 2019. Guidelines for the use of this product are described in the ICON Rules of the Road (https://icon.ssl.berkeley.edu/Data)

Responsibility for the mission science falls to the Principal Investigator, Dr. Thomas Immel at UC Berkeley: Immel, T.J., England, S.L., Mende, S.B. et al. Space Sci Rev (2018) 214: 13. https://doi.org/10.1007/s11214-017-0449-2

Responsibility for the validation of the L1 data products falls to the instrument lead investigators/scientists.

- \* EUV: Dr. Eric Korpela : https://doi.org/10.1007/s11214-017-0384-2
- \* FUV: Dr. Harald Frey : https://doi.org/10.1007/s11214-017-0386-0
- \* MIGHTI: Dr. Christoph Englert : https://doi.org/10.1007/s11214-017-0358-4, and
- https://doi.org/10.1007/s11214-017-0374-4
- \* IVM: Dr. Roderick Heelis : https://doi.org/10.1007/s11214-017-0383-3

Responsibility for the validation of the L2 data products falls to those scientists responsible for those products.

- \* Daytime O and N2 profiles: Dr. Andrew Stephan : https://doi.org/10.1007/s11214-018-0477-6
- \* Daytime (EUV) O+ profiles: Dr. Andrew Stephan : https://doi.org/10.1007/s11214-017-0385-1
- \* Nighttime (FUV) O+ profiles: Dr. Farzad Kamalabadi : https://doi.org/10.1007/s11214-018-0502-9
- \* Neutral Wind profiles: Dr. Jonathan Makela : https://doi.org/10.1007/s11214-017-0359-3
- \* Neutral Temperature profiles: Dr. Christoph Englert : https://doi.org/10.1007/s11214-017-0434-9
- \* Ion Velocity Measurements : Dr. Russell Stoneback : https://doi.org/10.1007/s11214-017-0383-3

Responsibility for Level 4 products falls to those scientists responsible for those products.

- \* Hough Modes : Dr. Chihoko Yamashita : https://doi.org/10.1007/s11214-017-0401-5
- \* TIEGCM : Dr. Astrid Maute : https://doi.org/10.1007/s11214-017-0330-3
- \* SAMI3 : Dr. Joseph Huba : https://doi.org/10.1007/s11214-017-0415-z

Pre-production versions of all above papers are available on the ICON website. http://icon.ssl.berkeley.edu/Publications

Overall validation of the products is overseen by the ICON Project Scientist, Dr. Scott England.

NASA oversight for all products is provided by the Mission Scientist, Dr. Jeffrey Klenzing.

Users of these data should contact and acknowledge the Principal Investigator Dr. Immel and the party directly responsible for the data product (noted above) and acknowledge NASA funding for the collection of the data used in the research with the following statement : "ICON is supported by NASA's Explorers Program through contracts NNG12FA45C and NNG12FA42I"

These data are openly available as described in the ICON Data Management Plan available on the ICON website (http://icon.ssl.berkeley.edu/Data).

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